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## INFRARED EMISSION FROM GAS-AEROSOL REACTIONS

by

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September 1982

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US ARMY ARMAMENT RESEARCH AND DEVELOPMENT COMMAND Chemical Systems Laboratory

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Infrared (IR) Infrared Emission Gaseous Amonia Sulfuric Acid-amonia IR Luminescence Gas-aerosol Reactions

IR Laser Exothermic Reactions Octanoic Acid-amonia

Octylamine-hydrogen Chloride Thermal Emission Acid-base Reactions Volatile Co-aerosol Chlorosulfonic Acid Aerosol Spectral Radiance

20. ABSTRACT (Continue on reverse side if necessary and identify by block number)

The present investigation has been designed to produce infrared emission by means of an exothermic reaction between a liquid aerosol and a gas. A number of gas-aerosol systems employing acid-base reactions have been examined and significant levels of radiation observed from the reactions of chlorosulfuric acid aerosol with gaseous amonia and water. Other systems which were screened including sulfuric acid-ammonia, octanoic acid-ammonia, and octylaminehydrogen chloride, have produced detectable levels of radiation. Some metnods

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	for the investigation of emission from the gas-aerosol reactions have been explored, and the results of these studies utilizing a chlorosulfonicacid aerosol system are presented.

#### PREFACE

The work reported herein was authorized by Contract DAAK11-80-C-0051, INFRARED EMISSION FROM GAS-AEROSOL REACTIONS. The work was performed from Department of Chemistry, Drexel University, Philadelphia, Pa. 19104

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I wish to acknowledge the contributions of Mr. K. Heaps, Ms. G.

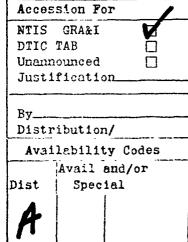
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#### I. INTRODUCTION

Infrared radiation in the atmosphere above normal background levels can be produced in a variety of ways. For example, combustion gases can produce significant amounts of radiation in the infrared region (2 to 20 m). However, the total mass of material, and thus the radiant emittance, is small. In addition, the gas cloud rapidly cools and disperses. In order to significantly increase the amount of airborne material, an aerosol must be employed.

At ambient temperatures (25°C) the maximum of the blackbody emission curve is at about 1030 cm<sup>-1</sup> (9.7µm), which means that a few degrees increase in temperature will produce an increase in spectral radiance on the order of 5% at 5 to 10µm. A highly conducting and absorbing substance may approximate a blackbody, but most real acrosol particles will at best be "grey" bodies with perhaps some superimposed structure (selectivity). The (equilibrium) thermal emittance of an acrosol particle will therefore be less than that of a blackbody at the same temperature, according to Kirchhoff's Law. A second and more selective method could involve the production of infrared fluorescence. This would have the advantage of being selective as to wavelength range and would thus also require much less total energy input. Unfortunately, due to the rapidity with which vibrational energy degrades, it does not seem possible to employ this approach. In other words, in order to obtain infrared (IR) luminescence in any observable yield, a dilute gas and a high energy IR laser is normally required. We have therefore focused attention on thermal

emission.

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In order to accomplish the desired goal, an exothermic chemical reaction must be employed to raise the temperature of the aerosol particle above ambient. This reaction must ultimately involve either the reaction of a gas with the aerosol or the simultaneous generation of a very highly disperse co-aerosol which will rapidly coagulate with the coarser aerosol and react. This is technically more difficult, and there would always be the problem of the highly disperse aerosol coagulating with itself. In the former case, the gas could be generated as a volatile co-aerosol.

A large number of chemical reaction studies have been performed on the upper end of the particle size spectrum with regard to the combustion of fuel sprays and dust clouds<sup>2</sup>. These involve particles in the super-micron range (10 to 1000µm). Chemical reactions responsible for heterogeneous nucleation have also been investigated, 3-5, and the particle sizes here are below 0.1µm. However, once aerosols are produced either by nucleation and growth in the atmosphere or by nebulization followed by settling and coagulation, most of these achieve a relatively stable existence in the 0.1 to 10µm range as smokes, fogs, etc.<sup>6,7</sup> These systems are polydisperse, with number densities generally less than 10<sup>7</sup>cm<sup>-3</sup>. Nonetheless, there have been relatively few studies of the reactions of gases with aerosol particles in this size regime.

In principle, the reaction rate may be controlled by gas phase diffusion, diffusion of reactants and/or products in the particle, by the bulk chemical reaction, or by processes occurring in the droplet interface region. Cadle and Robbins<sup>8</sup> have developed equations for some limiting cases and applied them to the reaction of ammonia with sulfuric acid aerosols in the 0.2 to 0.9 m range, and to the reaction of  $NO_2$  with a sodium chloride aerosol. In the former case, it was suggested that the rates in concentrated  $H_2SO_4$ 

droplets, the rate was too fast to measure on their apparatus, and it was presumed that the rate was controlled by gas phase diffusion. $^{8,9}$ .

There have also been a number of studies of the metal-catalyzed oxidation of  $SO_2$  in aqueous aerosols. Johnstone and Coughanour  $^{10}$ using a suspended 500 to 1000 $\mu$ m drop, concluded that the reaction was liquid-phase controlled with all of the reaction occurring in the outer shell at high manganese concentration. At lower catalyst concentration, the  $SO_2$  penetrated to the center of the drop. The oxidation of  $SO_2$  in 100 to 1000 $\mu$  aqueous ammonia drops was also controlled by liquid phase diffusion. However, extrapolation of the manganese catalyzed oxidation rates in the supermicron drop to an atmospheric fog of 20 $\mu$ m droplets predicted a rate 500 times that observed.  $^{12}$ 

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Because of the importance of atmospheric reactions, <sup>13</sup> studies have focused on water droplets and reactions involving SO<sub>2</sub>, NO<sub>2</sub>, NH<sub>3</sub>, H<sub>2</sub>SO<sub>4</sub>, and various metal salts. In these cases, the frequently complex reactions are generally also accompanied by droplet growth. This is not the case in the well-controlled reaction of submicron sized 1-octadecene droplets with bromine gas. <sup>14</sup> However, this reaction is slow and not highly exothermic.

Nonetheless, these studies do indicate that it should be possible to produce thermal emission as a result of a gas-aerosol reaction, and we report here the results of initial investigations utilizing exothermic reactions between gases and liquid aerosols.

#### 2. RESULTS AND DISCUSSION

## 2.1 Aerosol Generation and Detection.

For these initial studies, the aerosols were generated by means of a

compressed air nebulizer (DeVilbis). This generator produces an aerosol with a wide distribution of sizes (i.e. - polydisperse) and with a mass median diameter generally in the range of 2 to  $4\mu m$ , depending upon the liquid. However, this generator is easy to employ, and the mass concentration of the aerosol is quite reproducible under the same conditions of flow rate and tank pressure. Nitrogen gas (20 psi) at a flow rate of 3 to 6 2 min was employed, resulting in a reproducibility of about 5% for sulfuric acid and dibuty1phthalate (DBP).

All of the studies reported here have been carried out in the reaction tubes shown in Figure A-1. The wall-less tube (Figure A-1a) will be referred to as the "open" tube, and the other (Figure A-lb) as the "closed" tube. In both tubes the reactive gas stream emerges from the inner tube, the aerosol from the next concentric tube, and the sheathing gas from the outer tube. The prinicpal modifications from earlier designs are the mixing of reactive gas with aerosol via small holes at right angles to the aerosol flow rather than through a glass frit and, for the closed tube, the use of a larger (2.5 cm diameter) radiometer viewing port. The reaction tube, radiometer and exhaust are mounted in a metal frame shown schematically in Figure A-2. The entire setup is in a hood, but the exhaust fans in the hood cannot be used since the vibrations and airflow patterns cause excessive interference with the radiometer. For the open tube, a configuration employing a 4 inch diameter exhaust hose and auxilliary fan located outside the hood has proved effective. Arrangements involving cone-shaped vacuum exhausts were not effective. The radiometer is on a traverse for accurate and reproducible positioning, and thermistor probes may be inserted in the gas stream at variable locations.

The radiometer is a Molectron pyroelectric radiometer model PR200 with a response time of 3 sec for  $20\mu W$  cm<sup>-2</sup> full scale, a 0.1 Sr field of view, and

a detector area of  $0.2 \text{ cm}^2$ . The closed tube and the radiometer are fitted with a KRS-5 window which has a cutoff at about  $50\mu$ . Thus, essentially all of the emitted radiation is collected. The thermistor probes (YSI series 400) have a time response of 4 to 5 sec and a tolerance for exchange of probes of  $0.1^{\circ}\text{C}$ . The meter currently used gives readings to within +  $0.3^{\circ}\text{C}$ .

The output of the generator is determined by collection for a period of time, normally 1 min, followed by analysis. The DBP aerosol is collected on a millipore filter  $(\sim 1\,\mu)$  and weighed. The acid aerosols are collected in an impinger containing about 100 ml of water and then titrated either coulometrically or with NaOH to a phenolphthalein endpoint. It is also possible, for continuous monitoring, to use a glass pH electrode.

The reactions and reaction conditions employed in this study are given in table 1. The reactions listed are meant only to show the stoichiometry and not to imply anything concerning the extent or sequence of reaction.

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## 2.2 Gas-Aerosol Reactions.

Using the closed reaction tube (figure 15), emission from all of the reaction systems listed in table 2 was observed. There is an initial rapid jump in emission upon reaction, followed by a slower increase leveling off after about five minutes. Upon cutoff of the reaction gas, the radiometer decrease took place with a half-life also on the order of five minutes. Since the linear flow velocity of the gas-aerosol stream is on the order of 40 cm  $\sec^{-1}$ , heating of the glass wall of the reaction vessel seemed to be contributing to the emission. This was confirmed by simultaneous radiometer (R) and temperature (T) measurements using the thermistor probe. A plot of R vs T is shown in figure A-3 for the closed tube (figure A-3a). The large

# A. Reactions (aerosol underlined)

1. 
$$\frac{\text{C1SO}_3\text{H} + \text{H}_2\text{O} (g) \rightarrow \text{H}_2\text{SO}_4 + \text{HC1} (g)}{\text{H}_2\text{SO}_4 + \text{2NH}_3 (g) \rightarrow \frac{\text{(NH}_4)_2\text{SO}_4}{\text{NH}_4\text{C1} (s)}}$$

2. 
$$H_2SO_4 + 2NH_3 (g) \rightarrow (NH_4)_2SO_4$$

3. RCOOH + NH<sub>3</sub> (g) 
$$\rightarrow$$
 RCOONH<sub>4</sub>

4. 
$$RNH_2 + HC1 (g) \rightarrow RNH_3C1$$

## B. Conditions

- 1. Flow rates (1/min):
  - a. sheathing gas, 0.75
  - b. reactive gas, 0.75
  - c. aerosol;  $C1SO_3H$ , 5.4; all others 3.4
- 2. Concentrations  $(g/m^3)$ :
  - a. NH3, 20
  - b. H<sub>2</sub>0, 25
  - c. HC1, 8
  - d. aerosol; DBP, 5.1;  $C1S0_3H$ , 8.0;  $H_2S0_4$ , 1.0

Aerosol ·	<b>Ga</b> s	· ΔH <sup>a</sup>	IR <sup>b</sup>
octanoic acid	: NH <sub>3</sub>	21	W
octylamine	NC1	30	wm
H <sub>2</sub> S0 <sub>4</sub>	NH <sub>3</sub>	65	**
- ,	NH3/H20	64	m
C150 <sub>3</sub> H	H <sub>2</sub> 0	0-40 <sup>c</sup> 96-114 <sup>c</sup>	Mš
,·	NH <sub>3</sub> /H <sub>2</sub> 0	96-114 <sup>C</sup>	S

- a. Estimated heats of reaction in kcal per mole of aerosol material
- b. Indicates infrared emission (w=weak, m=medium, s=strong).
- c. Upper figure includes heat of hydration of reaction products.

hysteresis is indicative of wall effects. Nonetheless, the closed tube is still quite useful for screening studies due to its higher sensitivity. For example, there seems to be a rough correlation between the level of infrared emission and the enthalpy of reaction (table 2). The ClSO<sub>3</sub>H-H<sub>2</sub>O reaction seems to be an exception, although more quantitative studies are required. These studies may be carried out using an open tube vessel (vide infra).

In order to help eliminate effects due to wall heating, an open tube was employed (figure A-1b). That this system is indeed superior to the closed tube in this regard is shown by the rapid response to the introduction and withdrawal of the reactant gas (figure A-4) and the lack of R  $\underline{vs}$  T hysteresis (figure A-3b). It should be noted that the flow of reactant gas carrier was not changed. The carrier gas could be led through the reactant gas supply or could bypass it. The open tube does suffer from the disadvantage of being more susceptible to external perturbation such as wind currents and vibrations. Note the noise levels in figure A-4. Of course, this was recorded with a 3 sec time constant, and a 100 sec time constant could be used to determine steady-state levels. However, the observed range of measured power per unit area (R) with the open tube is about 1 to 10  $\mu$  cm<sup>-2</sup> compared with current "background" levels of about  $1 \mu$  cm<sup>-2</sup>.

#### 2.3 Temperature-Distance Profiles.

Since it is convenient to determine the fraction of  $C1S0_3H$  aerosol reacted in the closed tube, initial measurements of the temperature as a function of distance from the mixing zone were made in this reaction tube. The results are shown in figure  $\Lambda$ -5. The temperature is measured by the thermistor with the aerosol flowing with and without reactant gas. The quantity  $\Delta T$  is the difference between these two temperatures. It should be noted that all flow rates are identical since the reactant gas carrier stream

is on but bypassing the aqueous ammonia reservoir. The temperature is constant up to about 10 cm, and then begins to decrease. Since the linear gas velocity is about 100 cm  $\sec^{-1}$ , this represents a time of 0.1 sec. However, the reaction is not complete at this point, since analysis of the amount of  $C1S0_3H$  aerosol reacted at a point about 4 cm further downstream indicates  $\leq$  85% reaction.

This analysis was performed by passing the gas-aerosol stream through a bed of molecular sieve which passes a majority of the aerosol but no reactant gas. The ClSO<sub>3</sub>H aerosol was collected by two impingers in series since previous experience with this material showed that a significant fraction escapes a single impinger, probably due to the formation of HCl. The aqueous solution in each impinger was titrated coulometrically, and followed by monitoring the pH with a glass electrode. The output of the pH meter was displayed on a strip-chart recorder. This result indicates that the reaction rate rapidly slows down. A similar type of behavior was observed by Cadle and Robbins<sup>8</sup> in the reaction of ammonia with concentrated sulfuric acid droplets (.2 to .9µm) and ascribed to the diffusion of reaction product in the particle.

Since wall heating effects are known to be present in the closed tube, similar experiments were performed in the open tube. To serve as a basis for comparison, the rate of cooling of a thermally heated dibutyl phthalate (DBP) aerosol was also measured. The results are shown in figure A-6. The ClSO<sub>3</sub>H aerosol appears to maintain temperature longer than the heated aerosol, which it should if reaction is occurring. However, the difference is not great. It might be expected that the gas and droplet temperature in the case of the heated DBP aerosol are the same. The question as to what temperature is actually being measured in the case of the reacting aerosol will be addressed

below.

A simple treatment of the rate of heat loss fom a flowing stream by conduction and radiation, neglecting the velocity profile in the stream, convection, and the difference in speed between the various layers, predicts an expoential decay of temperature with distance. A plot of log  $\Delta T$  vs x does appear linear for the heated DBP aerosol as shown in figure A-7.

#### 2.4 Emittance-Temperature Profiles.

Over a range of temperatures, the (net) radiant emittance ( $\Delta R$ ) should be proportional to  $\Delta T$ . Therefore, a number of measurements of  $\Delta R$  vs  $\Delta T$  were made using heated DBP aerosol, heated gas, and the ClSO<sub>3</sub>H aerosol reaction system. The radiometer response for a heated DBP aerosol is shown in figure A-8. The time lag is greater than for the reaction system due to heating and cooling in the glass reaction tube. It should be noted that when the thermistor probe is in the radiometer field of view, considerably higher readings are obtained.

The radiant emittance-temperature profiles, obtained using the open tube and measured all at a fixed distance from the mixing zone, are shown in figure A-9. The solid line is a least-squares fit of the heated DBP data through the origin, and has a slope  $(\Delta R/\Delta T)$  of 0.56 $\mu$ W cm<sup>-2</sup>K<sup>-1</sup>. This is approximately 5% of blackbody. As expected because the aerosol provides only a small fraction of the mass of the emitting radiation all of which (aerosol and gas) is at the same temperature, heated gas (no aerosol) gives essentially the same result, albeit with slightly more scatter. The ClSO<sub>3</sub>H aerosol reaction data, on the other hand, exhibits a great deal of scatter. The corresponding values of  $\Delta R/\Delta T$  in figure A-9 are 0.82 (reactant gas on-off) and 0.36 (aerosol on-off)  $\mu$  W cm<sup>-2</sup>K<sup>-1</sup>. It was earlier reported, based on a few preliminary measurements, that  $\Delta R/\Delta T$  was higher for the reaction system  $^{15}$  but there is no definitive evidence for this in the more extensive set of data of figure A-9. We assume that for the same reaction conditions (flow rate, concentration, etc.) that the  $\Lambda$ R's will not depend on which reactant is controlled (on-off), therefore this variation in  $\Delta R/\Delta T$  must be attributed to variation in the temperature sensed by the thermistor. This raises the question as to the temperature actually being measured by the thermistor in the case of the gas-aerosol reaction. Where unlike the case of uniform heating of an inert aeosol flow in the oven, the gas temperature may differ from that of the reacting aerosol particles it carries. The rate of the gasaerosol reaction will likely be liquid phase controlled; i.e., controlled either by the rate of diffusion in the liquid drop or by the bulk reaction rate. Typically, this will involve reaction rates  $V_S$  or 3 X  $10^{-7}$  - 3 X  $10^{-8}$ moles  $cm^{-2}s^{-1}$  for 1  $\mu$  drops, although faster rates are possible. Under a sirale heat transfer model the aerosol will cool rapidly by conduction to the surrounding air such that a steady-state between the rate of heat generation

by reaction and heat loss by conduction should be rapidly ( $^{\circ}30\mu s$ ) achieved. Thus, the temperature of the aerosol drops ( $T_d$ ) and surrounding air in the aerosol cloud ( $T_c$ ) should be given by

$$T_{d} = T_{c} + rV_{s} H_{s}/\eta_{g}. \tag{1}$$

where r,  $H_s$  and  $\Pi_g$  are the drop radius, heat of reaction, and thermal conductivity of the air, respectively. For the above conditions and a  $\Delta H_s$  of 100 kcal mole<sup>-1</sup>,  $\Delta T = T_d - T_c$  should only be a few hundredths of a degree. If all of the heat generated warmed up the air mass with no loss, the cloud temperature would rise by about 12°C. However, the actual air temperature will depend upon the rate at which the "cloud" cools in our experimental apparatus. Using the result from figure A-7, maximum air temperatures of about 6°C are estimated. Now, aerosol will collect and react on the probe, some fraction of the heat will be transported away by conduction and radiation, and the remainder will heat up the probe to a temperature higher than the carrier gas temperature.

We have observed drops collected on glass slides under the same reaction flow and concentration conditions do show multiple impingements and "puddling" under the microscope. In addition, it is probable that the drops, impinging on the probe surface, expose more surface area and provide a stationary liquid being exposed to a more rapid, flowing supply of reactant gas. This will lead to a more rapid and complete reaction, heating the probe above the temperature of the gas. In this case the true  $\Delta T$  for the gas is less than that indicated by the probe, therefore the true value of  $\Delta R/\Delta T$  for the CISO<sub>3</sub>H aerosol stream is higher than indicated in the measurements. Since the measured values were comparable to the DBP result, the  $\Delta R/\Delta T$  for reacting CISO<sub>3</sub>H aerosol is greater than  $\Delta R/\Delta T$  for inert aerosol. For this to occur the drops themselves must be emitting more intensely than the DBP drops which were

known to be at the same temperature as the gas. Consequently this would mean that the drop and cloud temperatures were significantly different, as opposed to the  $^\Delta T$  =  $T_d$  -  $T_c$  of a few hundredths of a degree estimated by equation (1). Therefore the simplified physical description leading to equation (1) is not adequate for treating the case of a reacting aerosol particle in this size range.

As expected due to emission from the solid object, values of  $\Delta R/\Delta T$  increase if the thermistor is moved to a position within the field of view of the radiometer. Furthermore,  $\Delta R$  vs  $\Delta T$  profiles obtained with the thermistor probe centered in the radiometer field of view gave the same value of  $\Delta R/\Delta T$  (1.25 $\mu$ W cm<sup>-2</sup>K<sup>-1</sup>) for all three cases with relatively little scatter, as shown in figure A-10. This shows that the measured  $\Delta T$  does accurately reflect the temperature of the probe itself and that the radiometer responds linearly to the probes greybody emission over the temperature range of these experiments.

#### 3. CONCLUSION

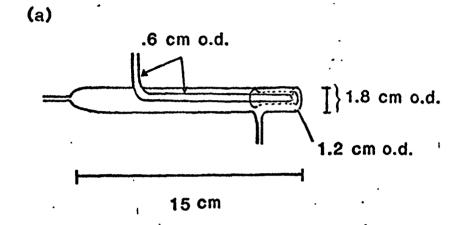
These initial studies have shown that it is possible to observe infrared emission from gas-aerosol reactions and to perform reasonably accurate measurements under controlled conditions. However, many questions remain to be answered before a quantitative model can be developed. In particular, is the drop and cloud (gas) temperature essentially the same or different? How must equation (1) be modified to properly reflect the heat transfer between a reacting drop and the surrounding gas? Future work must involve a study of rate of reaction, heat transfer and radiative emission, as well as a determination of the spectral distribution of the emitted radiation.

#### REFERENCES

- 1. "Handbook of Infrared Radiation From Combustion Gases," R. Goulard and J.A.L. Thomson, Eds., Washington Science and Technical Information Office, NASA (1973).
- 2. R.H. Essenhigh and I Fells, Discuss. Faraday Soc., 30, 208 (1960).
- P. Mirabel and J.L. Katz, J. Chem. Phys., 60, 1138 (1974).
- 4. C.J. Shen and G.S. Springer, Atmos. Environ., 10, 255 (1976).
- 5. H. Reiss, D.I. Margolese and F.J. Schelling, <u>J. Colloid and Interface Sci.</u>, <u>56</u>, 511 (1976).
- 6. J.P. Friend, Tellus, 18, 465 (1966).
- 7. M.J. Heard and P.D. Wiffen, Atmos. Environ., 3, 337 (1969).
- 8. R.D. Cadle and R.C. Robbins, Discuss. Faraday Soc., 301, 155 (1960).
- 9. R.C. Robbins and R.D. Cadle, <u>J. Phys. Chem.</u>, <u>62</u>, 469 (1958).
- 10. H.F. Johnstone and D.R. Coughanour, Ind. Eng. Chem., 50, 1169 (1958.
- 11. A.P. Van den Heuval and B.J. Mason, Quart. J. Roy Meteorol. Soc., 89, 271 (1963).
- 12. M.D. Carabi, Chem. Soc. Rev., 1, 411 (1972).
- 13. R.D. Cadle, J. Colloid and Interface Sci., 39, 25 (1972).
- 14. D. McRae, E. Matijevic and E.J. Davis, J. Colloid and Interface Sci., 53, 411 (1975).
- 15. R.A. Mackay, "Infrared Emission by an Aerosol Cloud," Final Report, U.S. Army Research Office, October 10, 1980.

# APPENDIX LIST OF FIGURES

- Figure A-1. Gas-aerosol Reaction Tubes.
- Figure A-2. Schematic Diagram of Reaction System.
- Figure A-3. Radiometer reading (R) vs probe temperature (T) for (a) closed tube and (b) open tube (vide text).
- Figure A-4. Radiometer response  $\underline{vs}$  time for the  $\mathrm{NH_3/H_20-C1S0_3H}$  gas-aerosol reaction system. The introduction and cutoff of reactant gas is denoted by arrows.
- Figure A-5. Temperature change ( $\Delta T$ ) <u>vs</u> distance for the C1SO<sub>3</sub>H-NH<sub>3</sub>/H<sub>2</sub>O reaction in the closed tube.
- Figure A-6. Temperature change ( $\Delta T$ ) <u>vs</u> distance for heated DBP aerosol (closed circles) and ClSO<sub>3</sub>H-NH<sub>3</sub>/H<sub>2</sub>O reaction (open circles) in the open reaction tube.
- Figure A-7. Log of the temperature change ( $\Delta T$ ) <u>vs</u> distance for the heated DBP aerosol in the open reaction tube.
- Figure A-8. Radiometer reading  $\underline{vs}$  time for a heated DBP aerosol in the open reaction tube. The maximum  $\Delta T$  is 6 C.
- Figure A-9. Net Radiometer reading ( $\Delta R$ ) vs temperature change ( $\Delta T$ ) for heated DBP (open circles), heated nitrogen (filled circles), and ClSO $_3$ H-NH $_3$ /H $_2$ O reaction (half-filled cirles). Either the reactant gas (left half-filled) or the aerosol (right half-filled) was turned on and off.
- Figure A-10. Net radiometer reading (ΔR) vs temperature change (ΔT) for heated DBP aerosol (open circles), heated nitrogen (filled circles), and ClSO<sub>3</sub>H-NH<sub>3</sub>/H<sub>2</sub>O reaction (half-filled) or the aerosol (right half-filled) was turned on and off. In all cases, the thermistor probe was in the radiometer field of view.



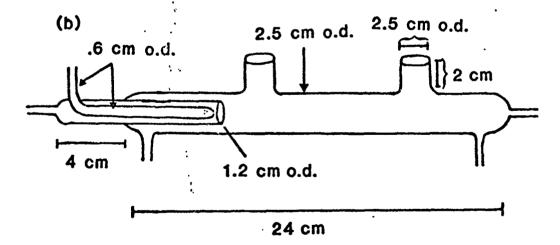


Figure A-1. Gas-aerosol Reaction Tubes

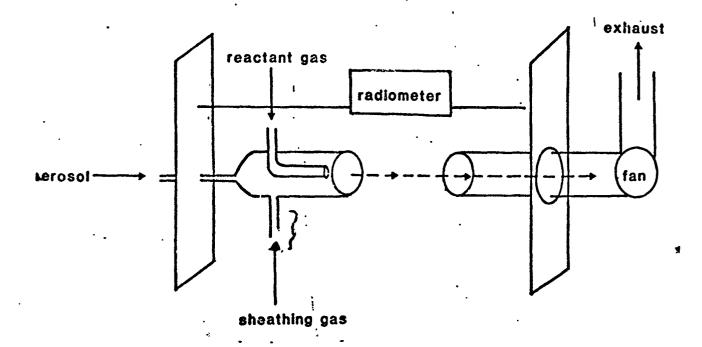


Figure A-2. Schematic Diagram of Reaction System

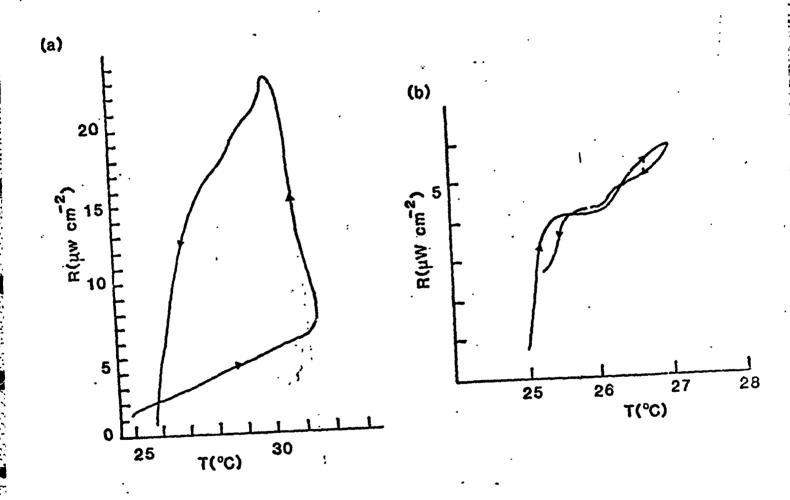


Figure A-3. Radiometer Reading (R) vs probe temperature (T) for (closed tube and (b) open tube (vide text).

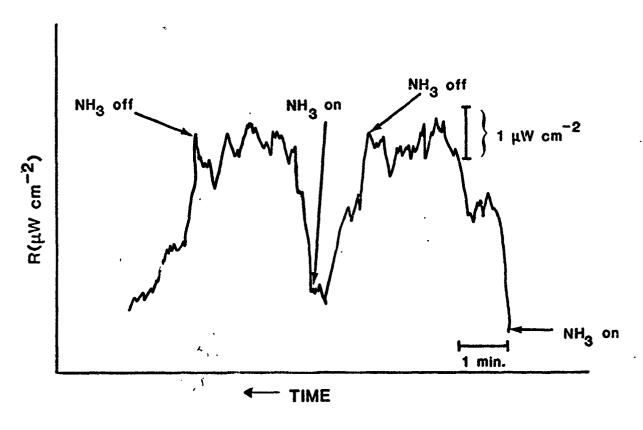


Figure A-4. Radiometer response <u>vs</u> time for the  $NH_3/H_2O-C1SO_3H$  gas-aerosol sea tion system. The introduction and cutoff of reactant was is denoted by arrows.

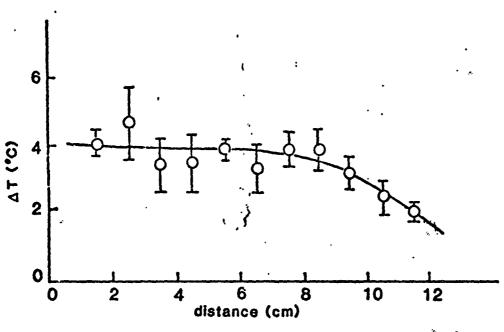


Figure A-5. Temperature change ( $\Delta T$ ) vs distance for the C1SO $_3$ H-NH $_3$ /H $_2$ O reaction in the closed tube.

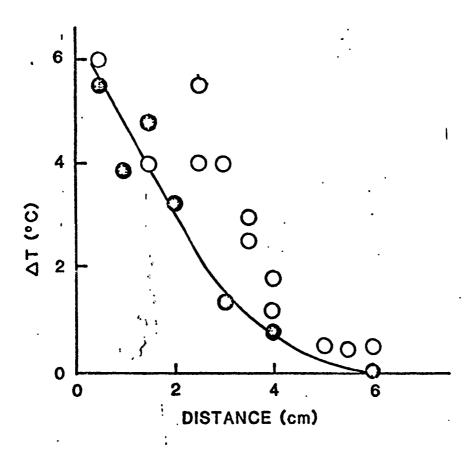


Figure A-6. Temperature change ( $\Delta T$ ) <u>vs</u> distance for heated DBP aerosol (closed circles) and ClSO<sub>3</sub>H-NH<sub>3</sub>/H<sub>2</sub>O reaction (open cirlces) in the open reaction tube.

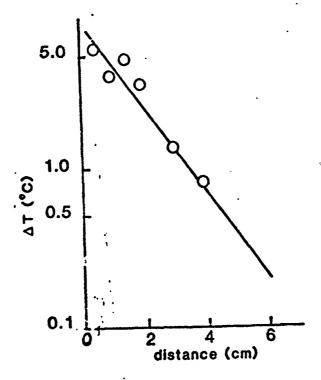


Figure A-7. Log of the temperature change ( $\Delta T$ ) <u>vs</u> distance for the heated DBP aerosol in the open reaction tube.

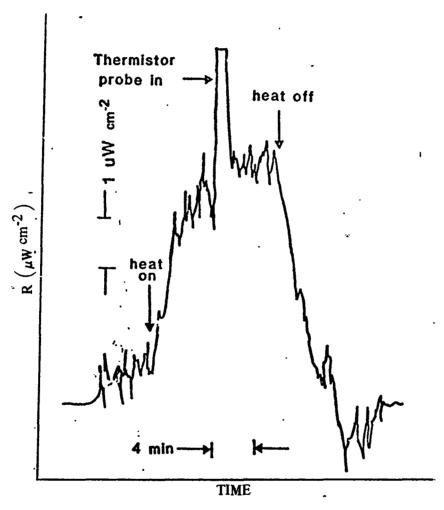


Figure A-8. Radiometer reading  $\underline{\mathbf{vs}}$  time for a heated DBP aeroscl in the open reaction tube. The maximum  $\Delta T$  is 6°C.

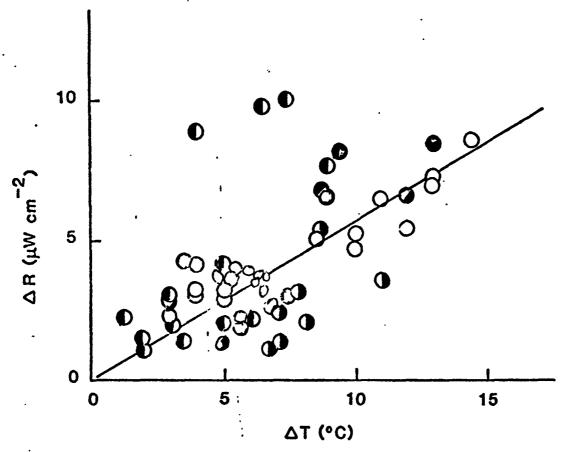


Figure A-9. Net Radiometer reading (ΔR) <u>vs</u> temperature change (ΔT) for heated DBP (open circles), heated nitrogen (filled circles), and ClSO<sub>3</sub>H-NH<sub>3</sub>/H<sub>2</sub>O reaction (half-filled circles). Either the reactant gas (left half-filled) or the aerosol (right half-filled) was turned on and off.

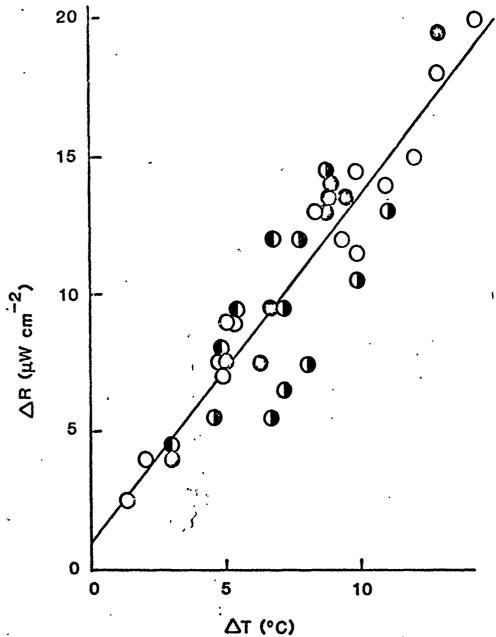


Figure A-10. Net radiometer reading ( $\Delta R$ ) <u>vs</u> temperature change ( $\Delta T$ ) for heated DBP aerosol (open circles), heated nitrogen (filled circles), and ClSO<sub>3</sub>H-NH<sub>3</sub>/H<sub>2</sub>O reaction (half-filled) or the aerosol (right half-filled) was turned on and off. In all cases, the thermistor probe was in the radiometer field of view.

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